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U.S. Department		
of Transportation		

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

orm Approved MB No.2120-0020	Electronic Tracking Numb	e

For FAA Use Only

Administration INSTRUCTIONS: Print or type all entries. See Title CFR 43.9, Part 43 Appendix B, and AC 43 9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 44701). Failure to report can result in a civil penalty for each such violation (49 U.S.C. 46301(a)). Nationality and Registration Mark Serial No. 1. Aircraft N6652K 929 Make Model Series Republic Seabee/Sky Enterprises RC-3 Address (As shown on registration certificate) Name (As shown on registration certificate) 2. Owner Fourth L.L.C. Address 1918 N 11TH Street City Muskogee State OK Zip 74401 3. For FAA Use Only The alteration/data identified herein complies with the applicable airworthiness requirements and is approved only for the above described aircraft subject to conformity inspection by a person authorized in FAR 43, section 43.7. DATE ET/30/2016 FAA INSPECTOR-SW FSDO, OKC 4. Type 5. Unit Identification Repair Alteration Unit Make Model Serial No. Х (As described in item 1 above) AIRFRAME **POWERPLANT** PROPELLER Type APPLIANCE Manufacturer 6. Conformity Statement A. Agency's Name and Address B. Kind of Agency Roger L. Kimball X U.S. Certificated Mechanic Manufacturer Address 3623 Club Estates Dr. #3 C. Certificate No. Foreign Certificated Mechanic City State OK Certificated Repair Station 002752096 Zip Country Certificated Maintenance Organization I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Extended range fuel per Signature/Date of Authorized Individual 14 CFR Part 43 App B 04/20/2010 7. Approval for Return To Service Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED FAA Fit. Standards Person Appoved by Canadian Manufacturer Maintenance Organization Inspector Department of Transport BY (Specify) FAA Designee Repair Station Inspection Authorization de uficate or Designation No Signature/Date of Authorized Individual 002752096 5-13-10

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished				
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration man				
The state of the s	rk and date work completed.)			
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N6652K	04/20/2010			
Notice of the second se				
Nationality and Registration Mark	Date			
This Form 337 is submitted for the purpose of recording and documenting an alteration that has been previously performed by persons, date and location unknown.				
The engine air intake filter has been replaced by a Brackett air filter assembly P/N BA-5810 of which a Brac	cket air filter element P/N BA-			
This filter has been approved for installation by STC-SA71GL on a Cessna T337H-SP which has a similar dimension/configuration air box and engine horsepower rating.				
The adequacy of filtering and air flow capability has been confirmed by Brackett Aero Filters, Inc. for this installation. See attached sheet for Air Flow data per Brackett Aero Filters, Inc.				
The air filter assembly (housing) has been modified by jectolling a ball, of				
The air filter assembly (housing) has been modified by installing a bale of the end for ease of removal from air box. Also installed on same end of housing is a brackett to secure filter assembly (housing) in air box. Both bale and securing bracket are attached by standard aircraft riets to housing. See attached photo.				
Modification of the air filter assembly bousing was accessed to the				
Modification of the air filter assembly housing was accomplished in accordance with standard practices per the air filter has been accomplished in accordance with the Installation Instructions of STC-SA71GL.				
Continued airworthiness shall be accomplished in accordance with the Instructions for Continued Airworthiness. SA71GL See attached sheet. Instructions for Continued Airworthiness.	ess directive of STC-			

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	8			
Additional Sheets Are Attached	Additional Sheets Are Attached			

N6652K

Instructions for Continued Airworthiness

This polyurethane air filter element has been designed to provide maximum dust collecting efficiency, non-restrictive air flow, and economical element replacement, while maintaining a lightweight filter. The element has been treated with a distinctive treatment called a wetted agent and is approximately 98% efficient. The wetted agent is an accompaniment in the efficient capturing of dust. In addition, the element has received a fire retardant treatment. For the above reasons, replace the element each 200 hours of use, 12 months or when 50% covered with foreign material. DO NOT WASH AND REUSE.

INSTALLATION INSTRUCTIONS

STEP 1. Remove filter element PN BA-5805 from bag.

STEP 2. Install into frame assembly and place grill PN 5804 over element.

STEP 3. Remove and discard existing filter.

STEP 4. Clean inlet area and inspect for irregularities.

STEP 5. Install filter assembly onto aircraft using the same procedure as original, using the

original hardware. Be sure screen and gasket side are facing downstream (engine side). See air flow arrows.

SERVICING: Under normal conditions, replace filter element, PN BA-5805, after 200 hours use

or 1 year intervals. Under severely dusty conditions, check daily and replace when element is 50% covered with foreign material.

INSPECTION INTERVALS: Pre-flight inspections, engine backfire inspection, 100 hour inspections, annual inspections, filter element replacements.

INSPECTION PROCEDURES

A. Pre-flight inspection: Per Pilots Operating Handbook, check filter assembly for security, damage or 50% contamination of element face. If found report to maintenance personnel prior to flight.

B. Engine start-up backfire inspection: Prior to flight, check the entire intake system for security or damage. If a fire was present, the downstream face of the foam element will show erosion. If any irregularities are found see Chart I and also refer to the Aircraft Maintenance Manual for the intake system.

C. At element replacement intervals: With the element removed, inspect the filter grill, filter frame, filter mountings and entire intake system for security, wear and any deformation. Note: On filter assemblies with gaskets, visually inspect inside and outside of frame for any signs of gasket looseness, movement or deterioration.