# US Department of Transportation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved	
OMB No. 2120-002	0

For FAA Use Only

Office Identification

Federal Aviation Administration

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

tor	each suc	ch violation (Sec	ction 901 Federal Avia	ation A	ct of	1958).	_		•	,			
1. Aircraft		Make Republic				Model RC-3							
	rcraft	Serial No.				Nationality and Registration Mark							
		1000				N5166B							
	Name (As shown on registration certificate				e) Address (As shown on registration certificate)								
2. Ov	wner	Davey Darrell L				6613 Santa Rosa Rd							
	<u> </u>	Davey Deborah A				Camarillo Ca 93012-5672							
	~					For FAA Use Only							
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				<del></del>	4.	Unit Identification			<u> </u>	5. Type	<u> </u>		
l	Jnit	Make			Model			Serial No.		Repair	Alteration		
AIRFRAME				cribed in Item 1 above)									
POWE	RPLANT												
PROPE	ELLER												
\PPLI/	ANCE	Type  Manufacturer											
				(	6. Co	nformity Statemen	t			<u> </u>			
4. Ag	ency's Na	ame and Addres	S			Kind of Agency		· · · · · · · · · · · · · · · · · · ·	C. Certif	ficate No.			
	K	enneth L.	Thompson		N	U.S. Certificated Mechanic			545767051				
		D Box 411	<del>_</del>		<u> </u>	Foreign Certificated Mechanic							
	Vi	ineburg, C	Ca. 95487			Certificated Repair Station  Manufacturer							
fı	certify th	at the repair and made in accord	i/or alteration made to dance with the require nd correct to the best	21111211115	s or ex	identified in item 4 a	above ar ederal A	nd described on the Niation Regulation	ne reverse ons and tha	or attachmer at the inform	nts hereto ation		
ate	9-	8-99			ł	nature of Authorize  1 Con Ta	_						
·	<u> </u>			7. Ap		al for Return To Se				<del></del>			
Purs Adm	uant to t	he authority given of the Federal A	ven persons specified Aviation Administration	i belov on and	w, the	e unit identified in APPROVED	item 4	was inspected in JECTED	the manr	ner prescribe	d by the		
<u>,                                    </u>	FAA	Flt. Standards ctor	Manufacturer		Insp	ection Authorization	1	Other (Specify)					
	FAA	Designee	Repair Station		Pers	on Approved by Tra ada Airworthiness G	nsport						
_	•	or Rejection	Certificate or Designation No.		Sign	nature of Authorized	d Individ	dual					
9-8-99 552273581				Douglas & Smith									

#### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

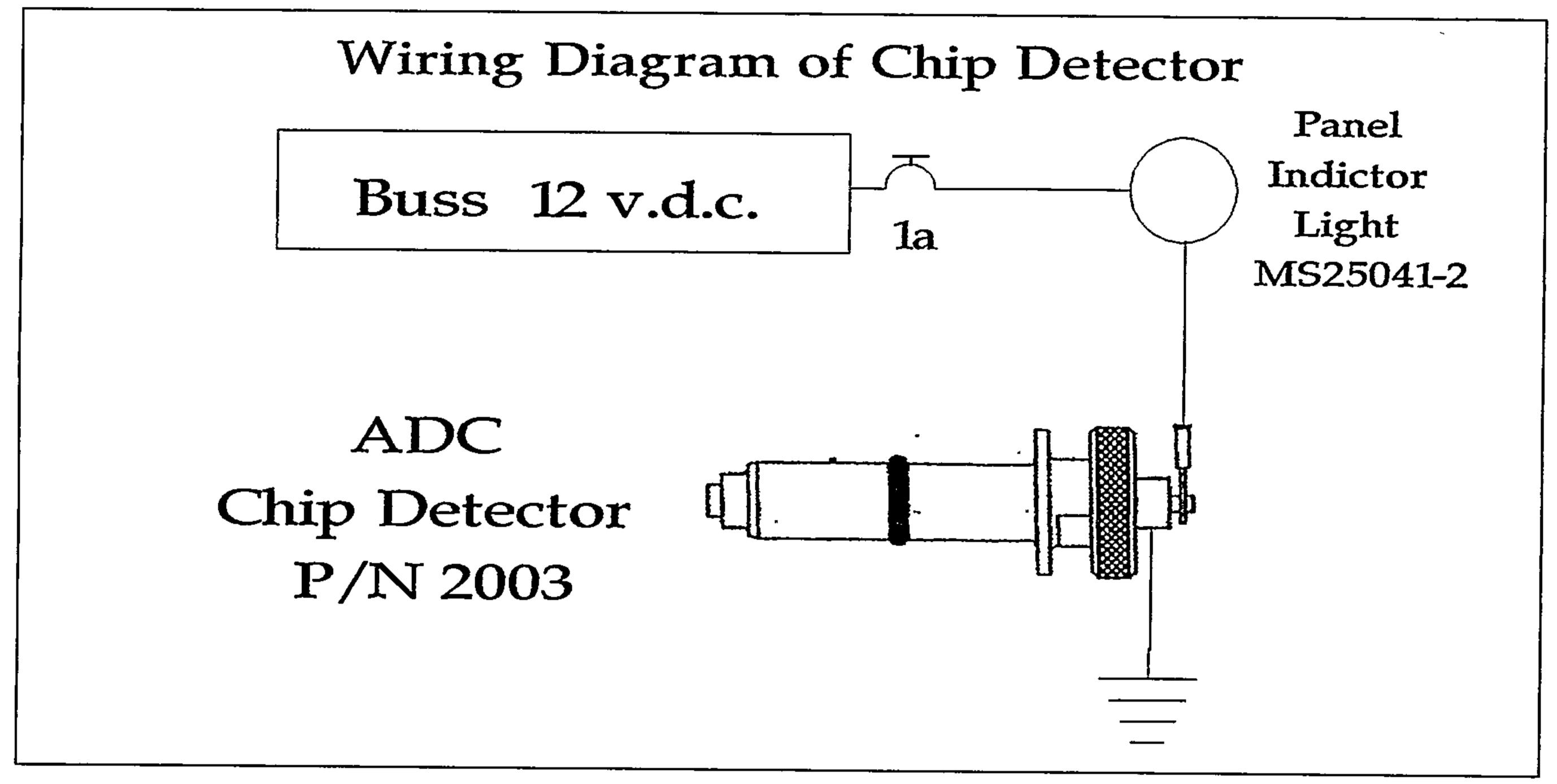
8. Description of Work Accomplished

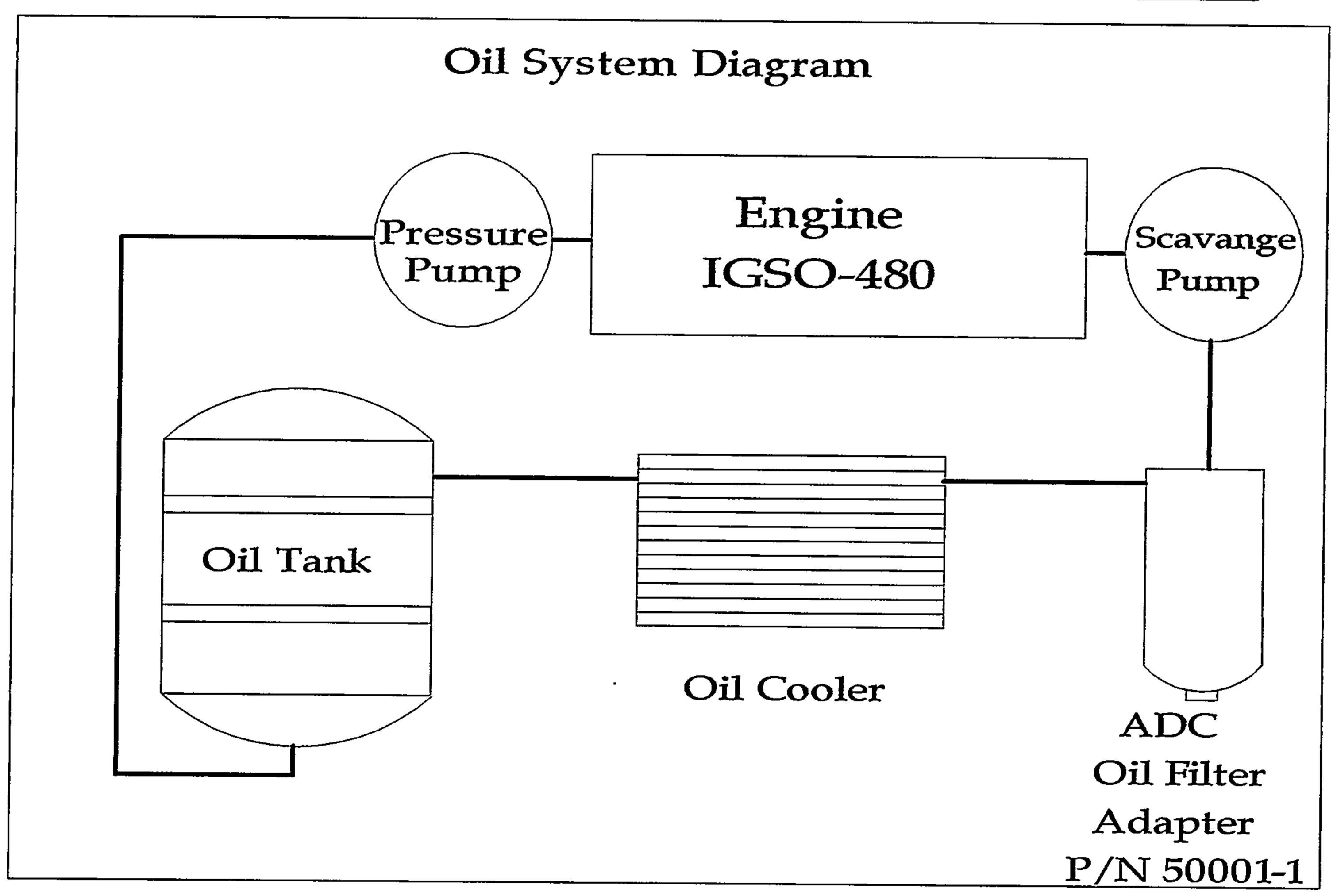
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Ser.# 1000 Republic RC-3 N5166B

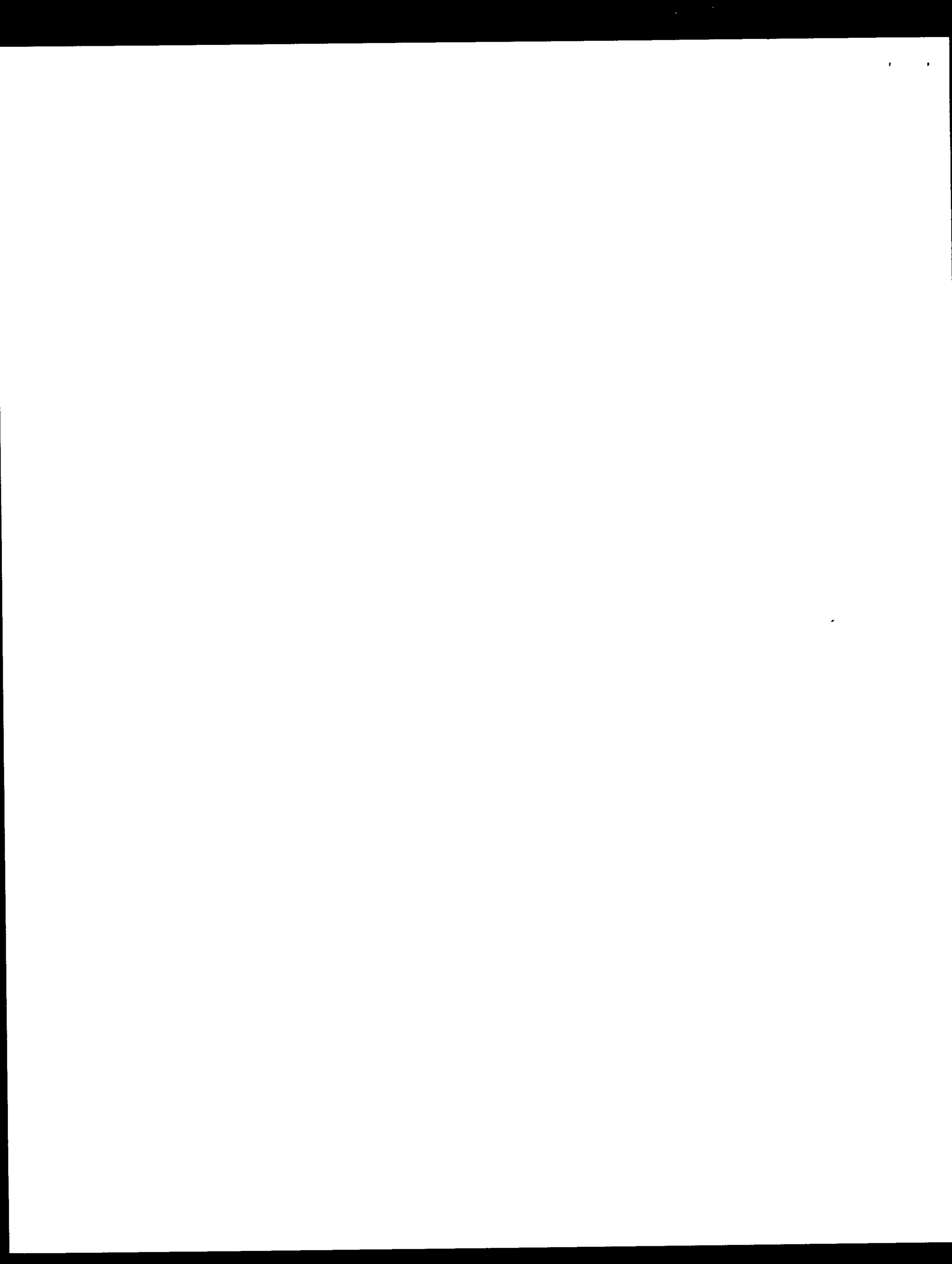
- 2. Description: Installed Aviation Development Corporation Spin-On Oil filter system P/N 07-06925 and chip detector P/N 2003 in left forward engine compartment. Oil filter adapter attached with AN3 bolts and AN363 nuts. Aeroquip 666-8 fluid hose TSO-C53a, Type D, with fire sleeve was used to connect the filter adapter to the engine and oil cooler. Chip indicator light P/N MS25041-2, push to test mounted in top center of instrument panel and uses a 1 amp circuit breaker for circuit protection. All work done in accordance with Aviation Development Corporation Report No. SP025, dated 10-17-96, ADC drawing# 50000 dated 9-2-96, 50010 dated 8-29-96, 50045 dated 10-14-97 (filter adapter), ADC Report No. CD30 dated 2-1-97, drawing# 1001 dated 3-4-94 (chip detector) and AC 43.13 1A, ch. 11, section 2, para.424, 429, section 3, para. 442, 443, 445 thru 451, fig. 11.7a, section 7, para. 514 thru 520.
- 3. Control, operation information: reference aircraft flight manual supplement
- 4. Servicing information: None
- 5. Maintenance instructions: Must be inspected annually in accordance with FAR 43 appendix D.
- 6. Trouble shooting information: N/A
- 7. Removal and replacement information: None
- 8. Diagrams: None
- 9. Special inspection requirements: None
- 10. Application of protective treatments: None
- 11. Data: None
- 12. List of special tools: None
- 13. For commuter category aircraft: N/A
- 14. Recommended overhaul periods: No additional overhaul time limitations
- 15. Airworthiness limitation section: No additional airworthiness limitations
- 16. Revision: A letter will be submitted to the local FSDO with a copy of the revised FAA form 337

## RC-3 N5166B S/N 1000 ADC Spin-On Oil Filter Adapter w/ Chip Detector





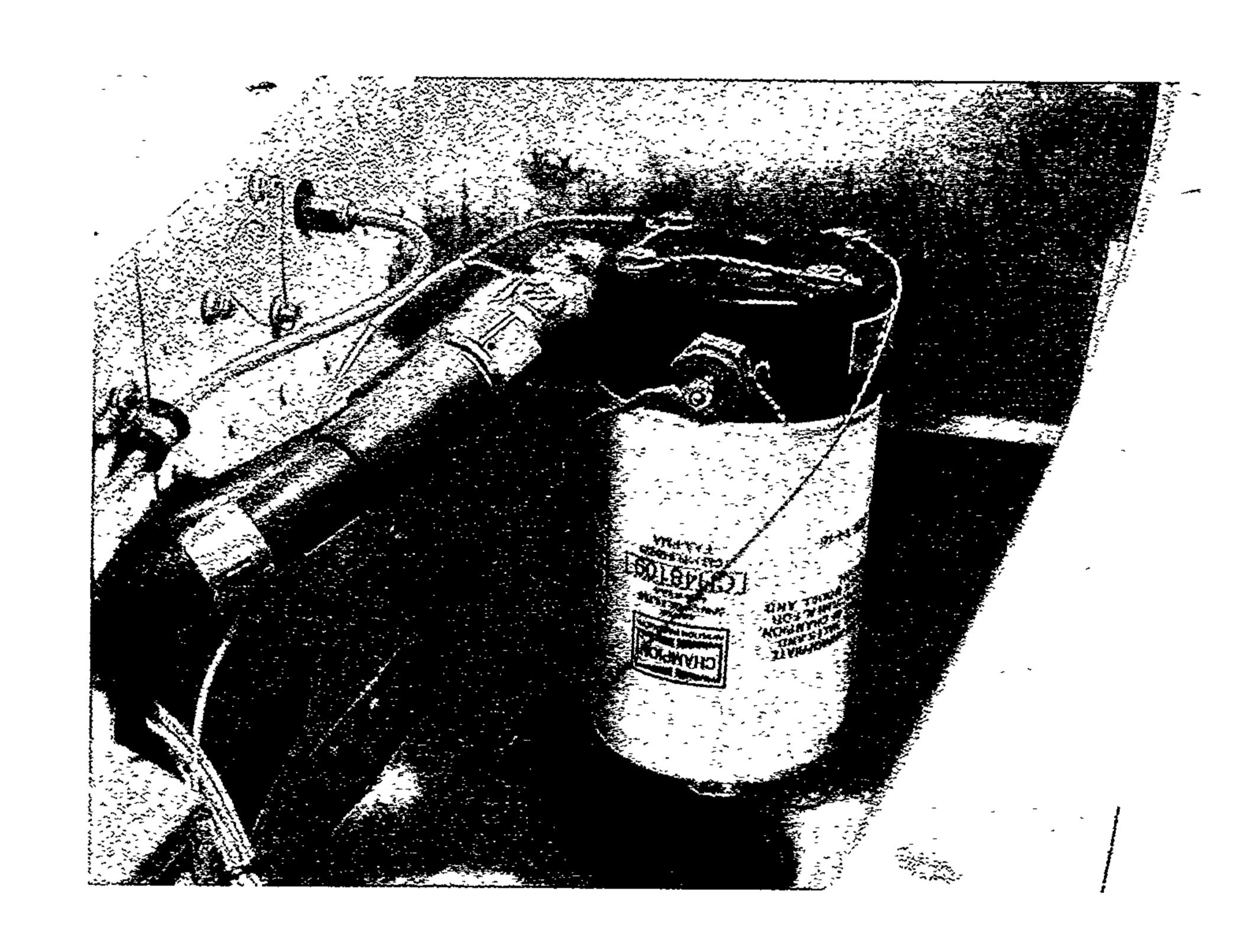
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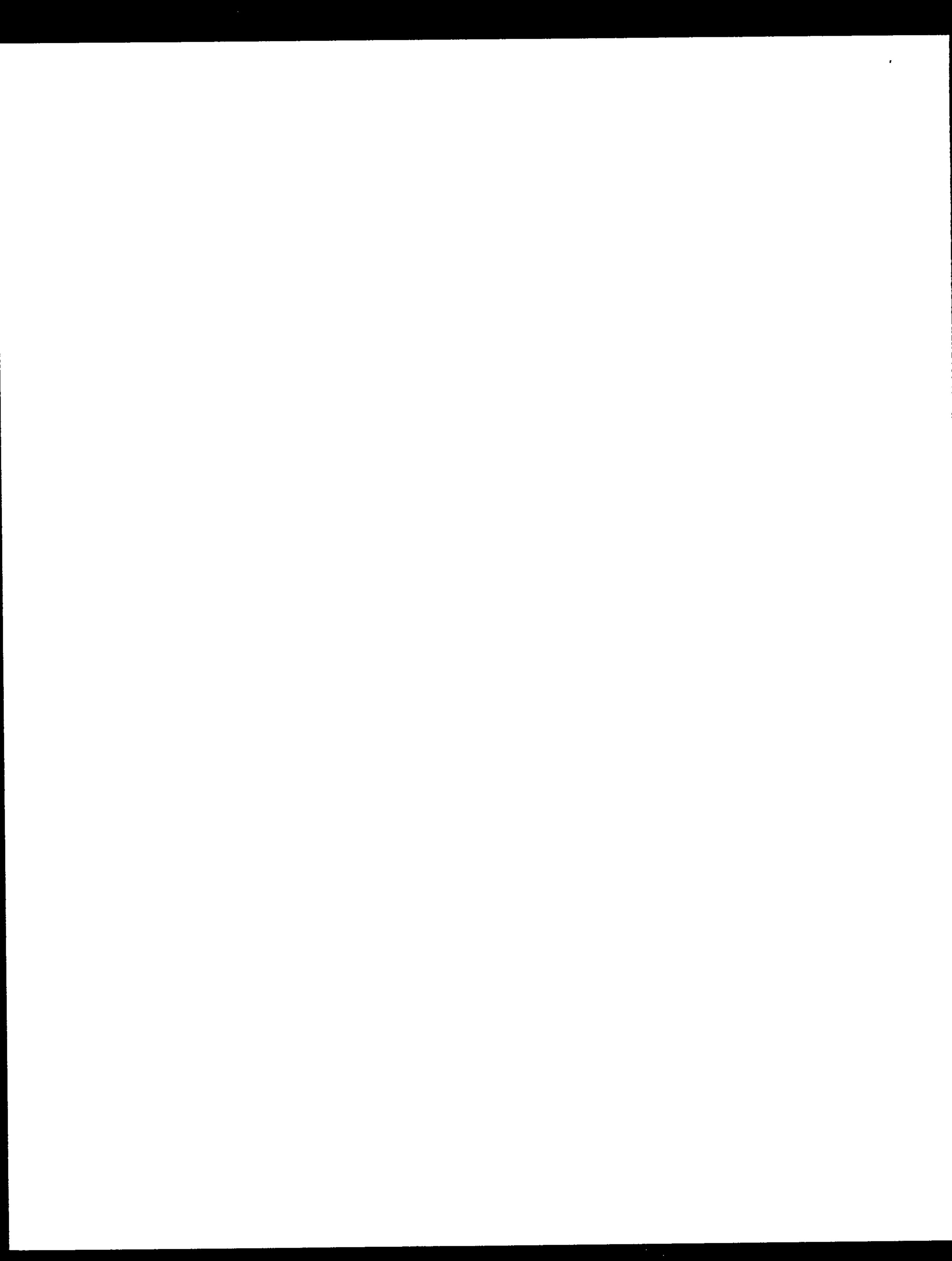
## RC-3 N5166B S/N 1000

ADC, Inc. Spin on Oil Filter and Chip Detector

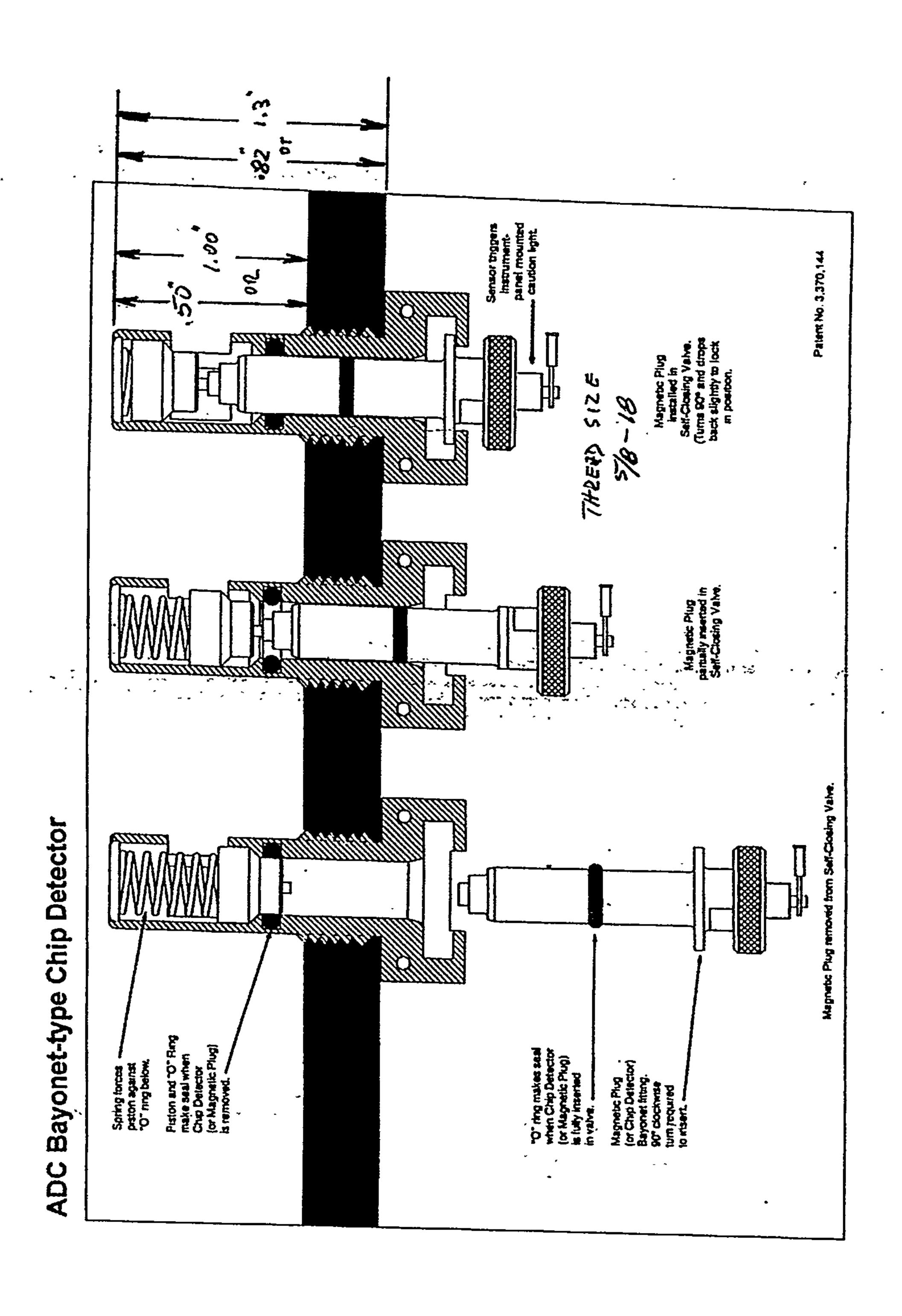
Located in forward left side of engine compartment



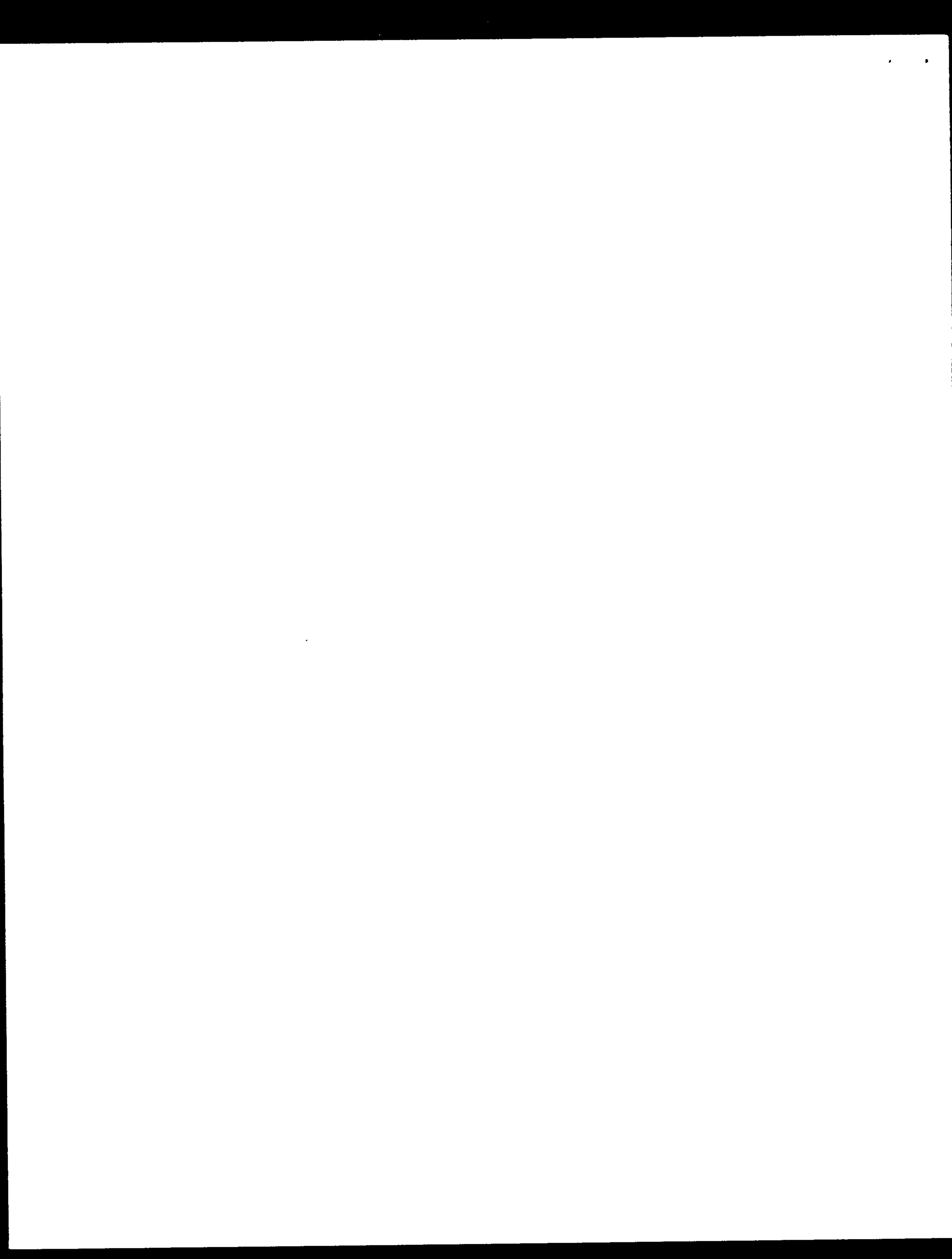
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# RC-3 N5166B S/N 1000 ADC Chip Detector Details



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### RC-3 N5166B S/N 1000

Aviation Development Corporation 1305 N.W. 200<sup>th</sup> Street Seattle, Washington 98177

#### Aircraft Flight Manual

Aircraft Type:	:_ <u>R</u>	public	RG3	<del></del>
Registration I	No:	NSI	66B	
Serial No:	100	20		

This supplement must be attached to Aircraft Flight Manual or Pilot's Operating Handbook when Aviation Development Corporation Chip Detector is installed.

The information contained in this document supplements or supersedes the basic manual only in those areas listed. For limitations, procedures and performance information not contained in this supplement, consult the basic airplane flight manual.

- I. LIMITATIONS No change.
- II. PROCEDURES if chip detector light comes on, land as soon as practicable. Remove and inspect chip detector for contamination. Clean and reinstall into chip detector housing.

Note: If an unusually high concentration of particles is present on chip detector magnet, this could be an indication of an abnormal engine wear condition, and further investigation should be conducted.

III. PERFORMANCE No change.

Date: 3/3/9/9

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